## ADVANCED FLUID DYNAMICS

Course Code: 19ME2201

#### **Prerequisites: Fluid Mechanics**

Course Outcomes: At the end of the course the student shall be able to

- CO1. Analyze and apply the concepts of turbulent flow to solve the fluid flow problems.
- CO2. Explain the concepts of boundary layer.
- CO3. Classify the compressible fluid flows and discuss stagnation properties.
- CO4. Solve nozzle, diffuser and shock wave problems of compressible fluids.
- CO5. Apply Prandtl, Rankine-Hugniot equations to solve oblique shock waves and discuss the Fanno curves.

### UNIT-I

Characteristics of turbulent flow - Reynolds equations of motion - turbulence modelling -Boussinesq Eddy viscosity concept - Prandtl's mixing length concept -Vonkaman similarity concept - Prandtl's universal velocity distribution-Karman - Prandtl velocity distribution power law for velocity in smooth pipes – Friction factor for smooth and rough pipes-Charts for friction factor in pipe flow.

Learning outcomes: At the end of this unit, the student will be able to

- 1. Explain the concepts of Boussinesq's eddy viscosity, Prandtl's mixing length and Vonkarman similarity concept. (L2)
- 2. Apply Prandtl's universal distribution equation to solve turbulent flow problems in pipes. (L3)
- 3. Use friction factor charts in pipe flows. (L3)

#### UNIT-II

Navier - Stokes Equations of motion - boundary layer over a flat plate - thickness of boundary layer -Prandtl's boundary layer equation - Vonkarmann momentum equation - shear stress and drag force - laminar boundary layer - turbulent boundary layer - pressure distribution in the boundary layer –boundary layer separation – drag and lift force – lift on an airfoil.

Learning outcomes: At the end of this unit, the student will be able to

- 1. Explain the concepts of boundary layer formation on a flat plate (L2)
- 2. Apply Von karman momentum equation to solve boundary layer problems (L3)
- 3. Discuss the methods of controlling boundary layer separation (L2)

#### **UNIT-III**

#### **10 Lectures**

Propagation of sound waves – Mach number – Mach angle – equation of sound wave. Energy equation – energy equation for non-flow and flow processes – adiabatic energy equation –

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#### **10 Lectures**

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stagnation enthalpy - stagnation temperature - stagnation pressure – stagnation velocity of sound – reference velocities – Bernoulli's equation – effect of Mach number on compressibility.

Learning outcomes: At the end of this unit, the student will be able to

- 1. Explain the propagation of sound waves in compressible fluid flow (L2)
- 2. Summarize the stagnation properties in compressible fluid flows(L2)
- 3. Discuss the effect of Mach number on compressibility (L2)

#### UNIT-I V

Comparison of isentropic and adiabatic processes – Mach Number variation - expansion in nozzles – compression in diffusers – stagnation and critical states – area ratio as a function of mach number – impulse function - mass flow rate, flow through nozzles - convergent nozzles – convergent-divergent nozzles – flow through diffusers. Development of a shock wave – rarefaction wave – governing equations, Prandtl-Meyer relation – Mach number downstream of the shock wave – static pressure ratio across the shock - temperature ratio across the shock – density ratio across the shock - stagnation pressure ratio across the shock

Learning outcomes: At the end of this unit, the student will be able to

- 1. Explain the expansion or compression of fluid flow through nozzle and diffusers (L2)
- 2. Describe the conditions for the development of shock waves (L2)
- 3. Derive Prandtl- Meyer relation of shock waves (L6)

#### UNIT-V

Nature of flow through oblique shock waves – fundamental relations - Prandtl's equation – Rankine-Hugoniot equation. The Fanno curves – Fanno flow equations – variation of flow parameters.

Learning outcomes: At the end of this unit, the student will be able to

- 1. Explain the formation of oblique shock waves (L2)
- 2. Derive Prandtl and Rankine Hugoniot equations for oblique shock waves (L6)
- 3. Illustrate the Fanno flow curves (L4)

#### **TEXT BOOKS:**

- 1. P. Balachandran, *Engineering Fluid Mechanics*, First Edition, PHI Learning Private Limited, New Delhi, 2012.
- 2. S.M. Yahya, *Fundamentals of Compressible Flow With Aircraft and Rocket Propulsion (SI UNITs)*, Fifth Edition, New Age International Publishers, NewDelhi, 2016.

#### **REFERENCE BOOKS:**

- 1. Yunus A. Cengel and John M. Cimbala, *Introduction to Fluid Mechanics*, Second Edition, Tata McGraw-Hill, 2010.
- 2. S.W. Yuan, Foundations of Fluid Mechanics, Prentice-Hall, 1970.
- 3. Patrick H. Oosthuizen and William E. Carscallen, *Compressible Fluid Flow*, First Edition,

McGraw-Hill Companies, Inc., New York, 1997.

# 10 Lectures

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